REVISION LIST

CHAPTER 4: FUEL SYSTEMS

The following list of revisions will allow you to update the Legacy construction manual chapter listed above.

Under the "Action" column, "R&R" directs you to remove and replace the pages affected by the revision. "Add" directs you to insert the pages shows and "R" to remove the pages.

PAGE(S) AFFECTED	REVISION # & DATE	ACTION	DESCRIPTION
4-1 through 4-3	0/02-15-02	None	Current Revision is Correct
4-4	1/09-18-02	R&R	Text Correction
4-5	TBA	R&R	Text and fig. 4:C:1 Correction
4-6	0/02-15-02	None	Current Revision is Correct
4-7	01/09-18-02	R&R	Text Correction
4-8	0/02-15-02	None	Current Revision is Correct
4-8	2/06-30-04	R&R	Added arrow for air flow direction.
4-9	2/06-30-04	R&R	Added photo.
4-10	2/06-30-04	R&R	Added photo.
4-1	3/12-15-04	R&R	New table of contents with page numbers and part nbr. change.
4-4	3/12-15-04	R&R	Part nbr. change.
4-11	3/12-15-04	R&R	Changed fuel pump fittings.
4-1, 4-4	5/05-15-07	R&R	Part number change only.
4-1, 4-2, 4-4, 4-10, 4-11	6/08-11-07	R&R	Part numbers changed.
4-8, 4-11	6/08-11-07	R&R	Change to instructions.
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LEGACY	4-1	FUEL SYSTEMS		
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Chapter 4: Fuel Systems

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1. INTRODUCTION

Starting at the filter cap, fuel finds its way through various drain holes through the wing tank. Fuel continues through one of the two check valves also called slosh doors. The slosh doors keep the fuel in the inboard bay during uncoordinated flight. Fuel continues through the fuel pickup and through the center wing section to the fuel selector valve. Next stop is the fuel pump, then the gascolator before it reaches the engine.

If you're using a Continental you must install a fuel return system as some fuel is returned back to the wing tanks.

2. PARTS LIST

#	PART NO. (P/N)	QTY	DESCRIPTION	OPTIONAL ITEM
				(not included with kit)
GEN	ERAL			
1)	SPRL-M-L	2	Fuel Caps, Metal	**Yes
2)	SPRL-KCI	2	Fuel Caps, Plastic	
FUE	LSELECTOR			
1)	4021	1	Fuel Selector Valve Mounting Flange	g
2)	4808	1	Fuel Selector, Continental	**Yes
3)	EFS 25-2-2B-M	1	Fuel Selector, Lycoming	**Yes
4)	K3000-3	4	Nutplates	
5)	MSC-34	8	Rivet, Cherry	
6)	101-0065 3K146		4 Sockethead	Screw
7)	AN525-832R10	4	Washerhead Screw	
FUE	LSUPPLYLINES			
1)	4805	2	Flexible Hose # 8 x 9.25'	
2)	4807		Pre-made line	**Yes
3)	4809		Pre-made line	**Yes
4)	165-0000/165-0001/165-0	0002		
	85000K83/84/85	10'	Antichafing Material for Fuel	Lines
5)	AN924-8D	2	Check Nut	
6)	MS21919-DG 8	4	Clamps	
7)	CS125-1032-12GCR	4	Click Bond Studs 3/16"	
8)	AN818-8D	8	Coupling	
9)	FU7-2	2	Fitting, with Screen	
10)	AN833-8D	2	Fitting, Elbow (Bulkhead)	
11)	AN832-8D	2	Fitting, Union (Bulkhead)	
12)	5052500 x .035	12'	1/2" Aluminum Tubing	
			for Fuel Lines	
13)	AN365-1032A	4	Nut, Nylock	
14)	AN819-8D	8	Sleeve	
15)	AN960-10L	4	Washer, Flat (thin)	
		N	Note:	
			Optional Parts available	through:
			(*) Lancair Avionics	_
			(**) Kit Components,	Inc.



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!	PART NO. (P/N)	QTY	DESCRIPTION	OPTIONAL ITEM
				(not included with kit
FUE	L VENT LINE CHECK VA	LVE		
1)	545	2	Check Valve	
FUE	L RETURN LINES			
1)	4806	2	Flexible Hose # 4 x 12'	
2)	5052250 x .035	18'	1/4" Aluminum Line	
3)	85000K83/84/85	10'	Antichafing Material for Fuel	Lines
4)	MS21919-DG 4	6	Clamps	
5)	CS125-1032-12GCR	6	Click Bond Studs 3/16"	
s)	AN818-4D	8	Coupling	
7)	AN 822-4D	2	Fitting, Elbow	
3)	AN833-4D	2	Fitting, Elbow (Bulkhead)	
9)	AN832-4D	2	Fitting, Union (Bulkhead)	
(10)	AN924-4D	4	Nut, Check	
11)	AN365-1032A	6	Nut, Nylock	
12)	AN819-4D	8	Sleeve	
13)	AN960-10L	6	Washer, Flat (thin)	
RHE	LPROBE (optional)			
1)	P-300-C-8	1	Fuel Probe (EI)	*Yes
2)	3010010	8'	Vision Microsystems	*Yes
ומוומ	I hoost DIIMD kit (ontion	al DIIDCHAS	SE ALL AS ONE WITH PUMP) KIT)
1)	5456-00-1	1	For 12V Lycoming 540	**Yes
2)	5455-00-1	1	For 12V Continental 550	**Yes
3)	145-0008-5416K32	1	Clamp	**Yes
4)	229-4-1	1	Fitting, Elbow	**Yes
5)	C5365 x 8	2	Fitting, 45 Degree	**Yes
7)	3814-6	1	Hose	**Yes
) 5)	22-4	1	Splice	**Yes
ומוזק	L DRAIN COVERS			
	F-391-18	4	Fuel Drain Valve	
1)	4271	4	Fuel Drain Fairing	
2)	44/I	4	ruci Diani raning	

Note:

Optional Parts available through:

(*) Lancair Avionics

(**) Kit Components, Inc.



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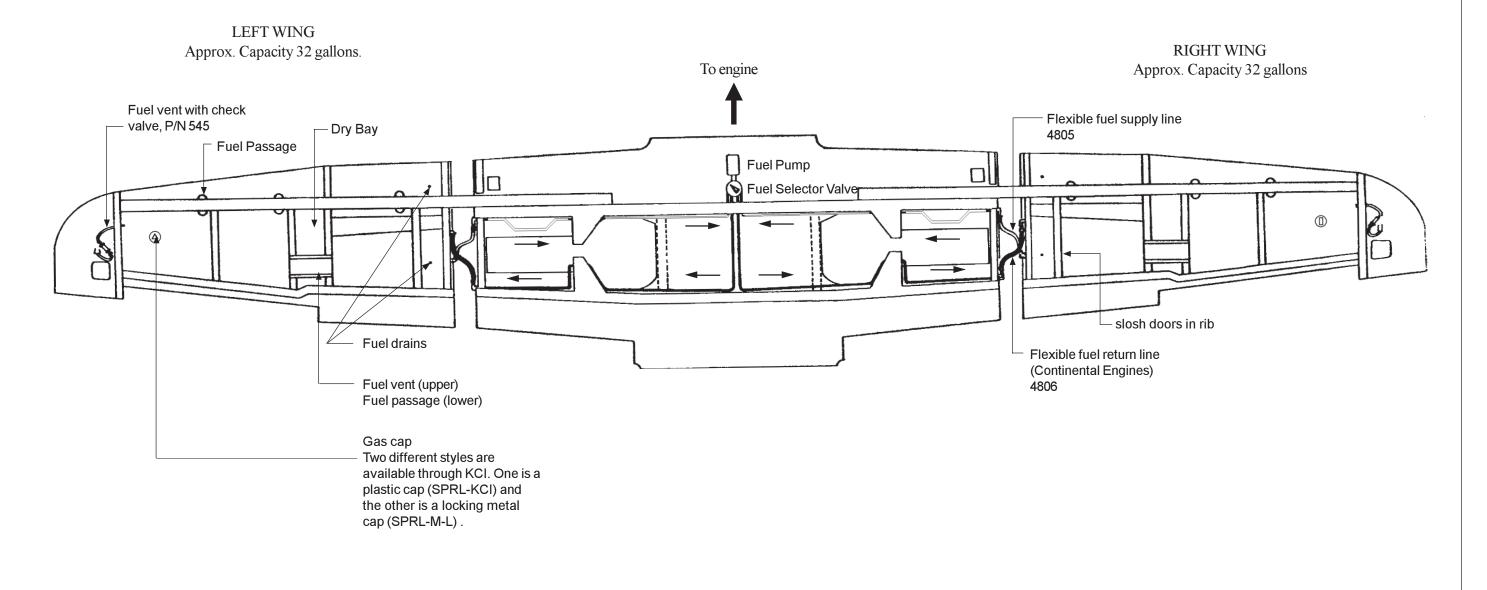
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3. CONSTRUCTION PROCEDURES

A. Legacy Fuel System - General Overview

Legacy Fuel System Fig. 4:A:1

CENTER WING SECTION



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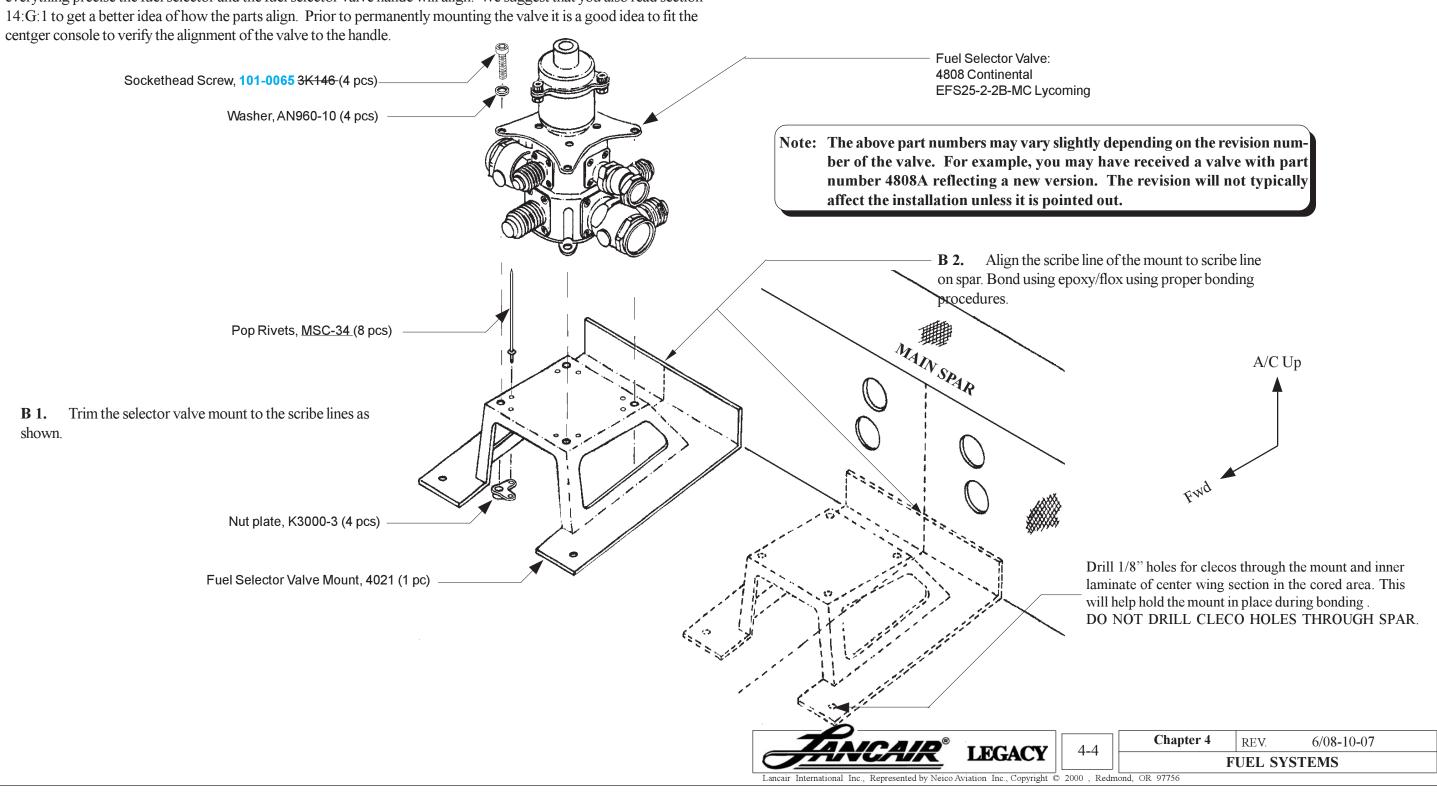
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B. Fuel Selector Valve (Optional)

Fuel Selector Mount Fig. 4:B:1

The fuel selector valve mounts just in front of the main spar along the fuselage center line. Note that the mount is slightly angled to make the valve parallel to the handle that mounts in the center console. Provided you build everything precise the fuel selector and the fuel selector valve hande will align. We suggest that you also read section 14:G:1 to get a better idea of how the parts align. Prior to permanently mounting the valve it is a good idea to fit the centger console to verify the alignment of the valve to the handle.



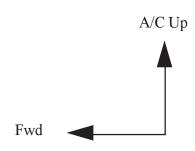
Fuel Supply Lines Fuel Supply Lines for Center WingSection (Exploded View) Fig. 4:C:1 LEFT OUTBOARD WING **SECTION** A/C Up 90° Bulkhead Fitting AN833-8D (2 pcs) Check Nut, AN924-8D (4 pcs) Shown with Coupling, AN818-8D and sleeve AN819-8D Fuel pickup of outboard wing section installed on each end. (pipe thread brass block, P/N FU3-2) 90° Fitting with screen FU7-2 (2 pcs) Straight Bulkhead Fitting AN832-8D (2 pcs) Shown with coupling, AN818-8D and sleeve AN819-8D installed on each end (connects to fuel selector valve) **CENTER WING SECTION** Enlarge hole at All fuel supply: scissor link rib as lines 5052-.500 x .035 required. sleeves AN819-8D The line must have a nuts AN818-8D bend in it similar to Inboard rib must be shown for wing flex. aligned to complete the WS-59 Rib supply line. See chapter 10 section B. Suggested passage hole for fuel probe wires. We suggest installing all fittings and in general, familiarize yourself with the system. Then complete the center wing section fuel system Secure with adel clamp. Access hole for the purpose of anytime after chapter 11. securing the fuel supply lines. Chapter 4 REV. 0/02-15-02

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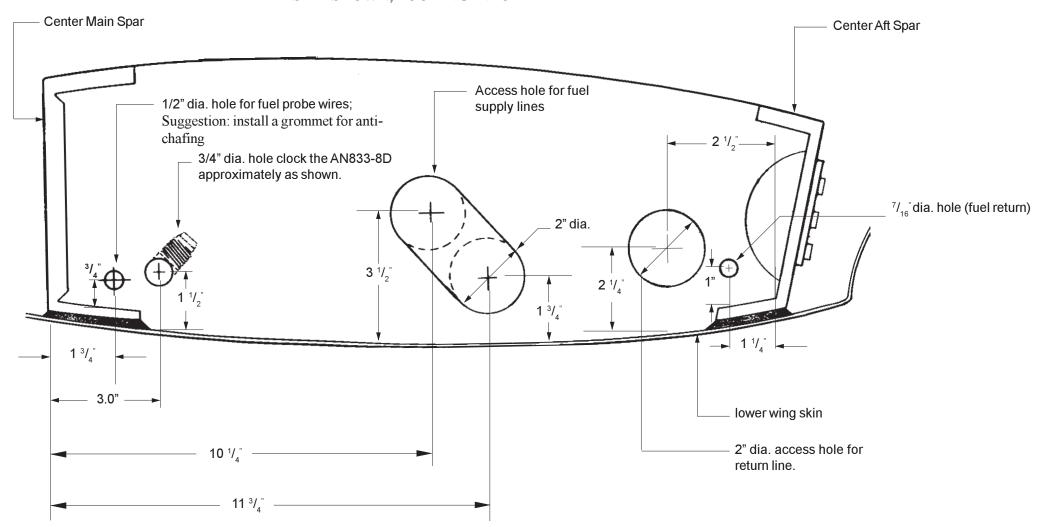
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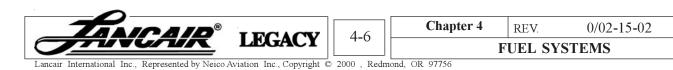
FUEL SYSTEMS

Fuel System Related Holes of WS 61.5 Rib Fig. 4:C:2

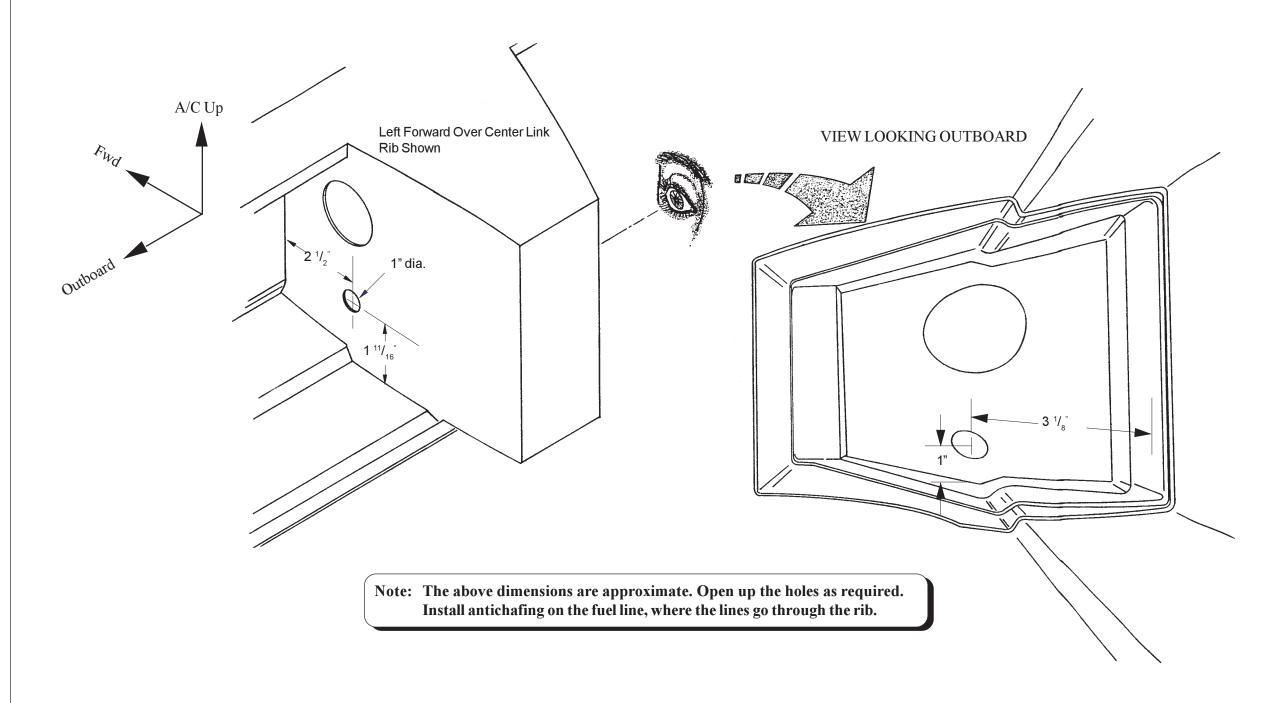


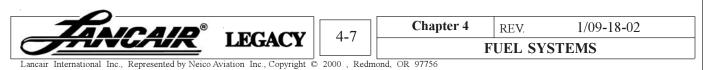
LEFT SIDE SHOWN, LOOKING INBOARD





Fuel Supply Lines through Over Center Link Rib Fig. 4:C:3

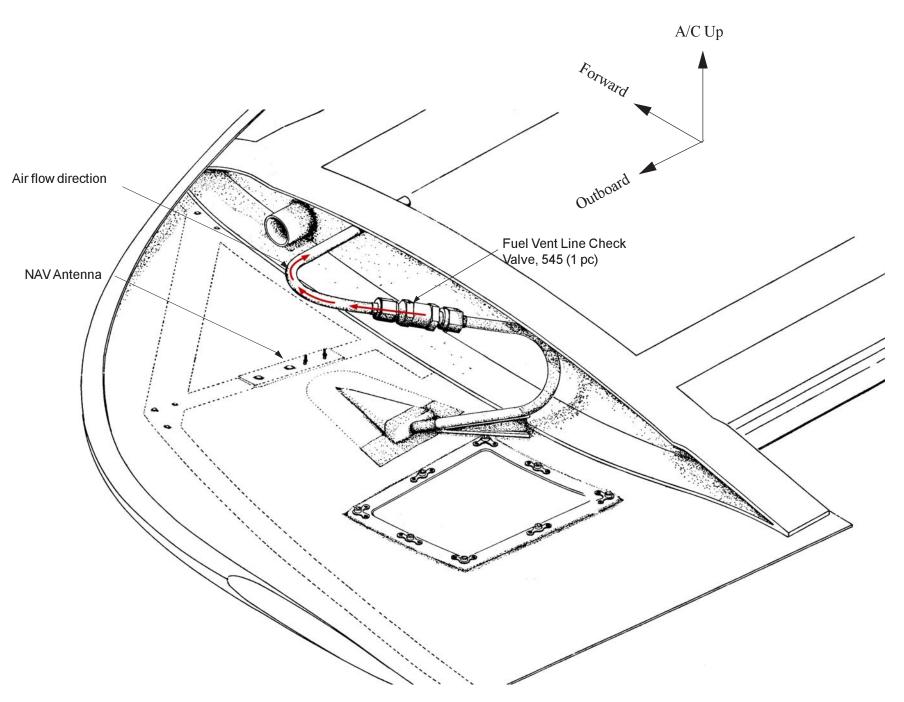




D. Fuel Vent Line Check Valve

D 1. Install the fuel vent line check valve as shown. The red arrow must point towards the tank, which the air flows into.

Installing Fuel Vent Line Check Valve Fig. 4:D:1



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Refer to this figure for the approximate routing of lines. The lines must clear the center console and the cross over weldment assembly. We suggest temporarily installing the center console and the cross over weldment. Refer to Figure 6:A:1 and Chapter 14. All return lines are 1/4" aluminum with AN819-4D Sleeves and AN818-4D Nuts Secure hard lines as required. Suggest

Use anti-chaffing material around holes as required. Part #165-000/165-0001/165-0002

Premade Hose, 4805

Bulkhead Fitting, AN833-4D

with AN924-4D Check Nuts
90° Bulkhead Fitting, AN833-4D
with Check Nuts, AN924-4D

45° Bulkhead Fitting, AN837-4D with Check Nuts, AN924-4D

The fuel lines and hydraulic lines must clear the flap mechanism. Temporarily mount the flap torque tube supports for reference.

Note: Aft spar not shown for clarity.



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locations for suports marked by an " *...

Also see Chapter 1.

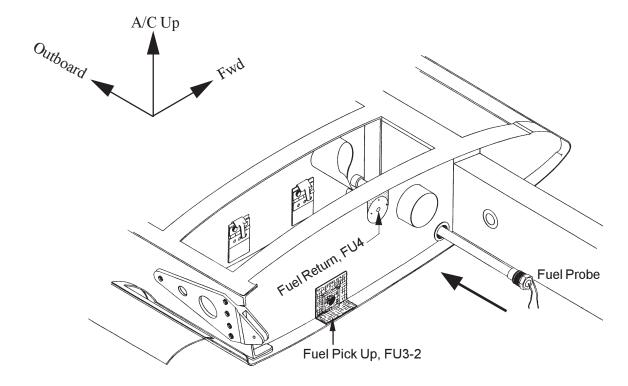
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F. Fuel Probe (Optional)

F1. Lancair offers two options for the fuel probes. One is for Vision Micro Systems (VMS) and the other is from Electronics International (EI). The installation is identical. Both systems are available through Lancair Avionics.

Warning: Apply antiseize on the threads of the aluminum fuel probe mount. Without antiseize the fuel probe will cold weld to the aluminum fuel probe.

Fuel Probe Installation Fig. 4:F:1

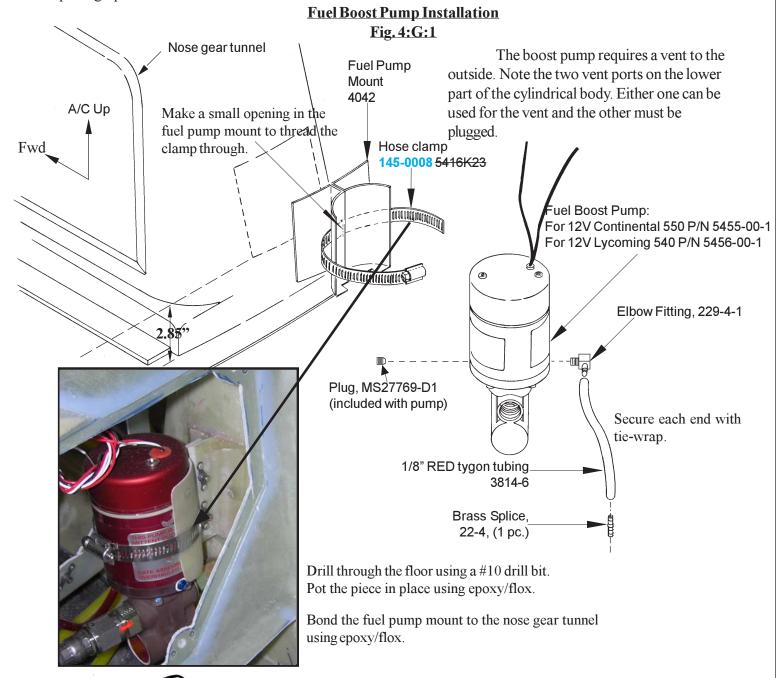


Fuel Probes VMS P/N: 3050010 Fuel Probes EI P/N: P-300-C-8

Note: The correct fuel probe length for the Lancair Legacy fuel tanks is 72".

G Fuel Boost Pump (Optional)

- **G1.** The type of fuel pump used depends on the type of engine installed. The Continental requires a two-stage fuel pump while the Lycoming a single stage. Two-stage means that the pump has a low boost. Refer to the pilot operating handbook for proper use.
- **G 2.** The new, longer fuel pump mount needs to be cut from web. Adjust the length of the mount as necessary. The piece holding the fuel pump can be either bonded or screwed to the other section of the mount. Refer to the photograph for current installation method.



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Refer to Blueprint #4862 for the location of the fuel supply and fuel return fittings in firewall. Fuel return is not needed with the Lycoming installations. This figure completes the fuel lines installation up to the firewall. **Fuel Line Installation** Fig. 4:G:2 Make a hole in the gusset as shown. Anti-chafe the hole. This is available through KCI, part number 165-0000/165-0001/165-0002 8500K83. Use anti-chaffing Sleeve, AN819-8D Nut, AN818-8D material around holes as required. Optional Pre-made Line, Main Spar Cross Over 4809 Weldment Center Spar Closeout Fuel Supply Line: Secure the fuel lines Fuel Supply Line Tubing Make a 90° bend and run **Engine Mount** using adel clamps 5052-500 x 035 outboard. Support with adel Gusset MS21919-DG8 and clamps. standoffs, CS125-1032-12**GCR**. Fuel Return Line: Route down to floor. Clear the cross over weldment and center console. Nose Gear Tunnel Fuel pump fittings: Fuel Supply Line, C5515x8 (1 pc) 90 deg. 5052 - .500 x 0.035 C5315x8 (1 pc) straight Sleeve. AN819-8D **NOTE:** Check that the fuel supply line Fuel Return Line, Nut, 5052 - .25 x 0.035 does not interfere with the floorboards. AN818-8D Sleeve, AN819-4D Nut, AN818-4D Optional Pre-made Line, 4807 Chapter 4 REV. 6/08-10-07 **LEGACY FUEL SYSTEMS** Lancair International Inc., Represented by Neico Aviation Inc., Copyright © 2000 , Redmond, OR 97756

H. Fuel Drain Covers

Fuel Drain Fairing Fig. 4:H:1

